



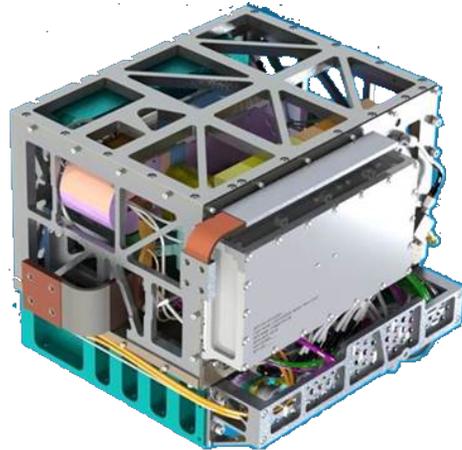
Discovery 2014 Briefing to Potential Proposers  
Capabilities and Considerations for Incorporation Onto a Discovery-Class Mission

Allen H. Farrington  
DSAC Project Manager  
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## DSAC is a Flight Mission – Demonstrate Clock on Orbit (TRL7)

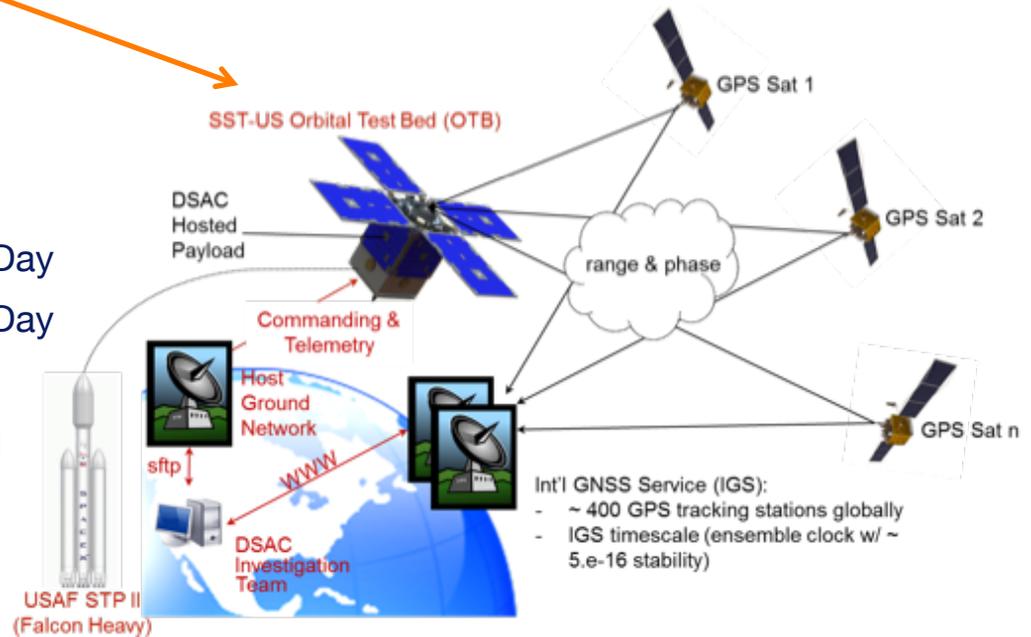


- Develop Advanced Prototype Clock for Year-Long Test Flight
- Demonstrate State of the Art Clock Stability (see below)
- Develop and Demonstrate Deep Space Navigation Tools and Usage of Clock in a One-Way Tracking Scenario

### Requirements:

Clock prior to launch:  $3.0 \text{ E-15 @ 1Day}$   
 On-orbit Clock Validation:  $2.0 \text{ E-14 @ 1Day}$

Current CBE predicts that DSAC in-space will actually meet ground based performance (A.D.  $< 3.0 \text{ E-15 @ 1 day}$ ) which outperforms any existing space clock.



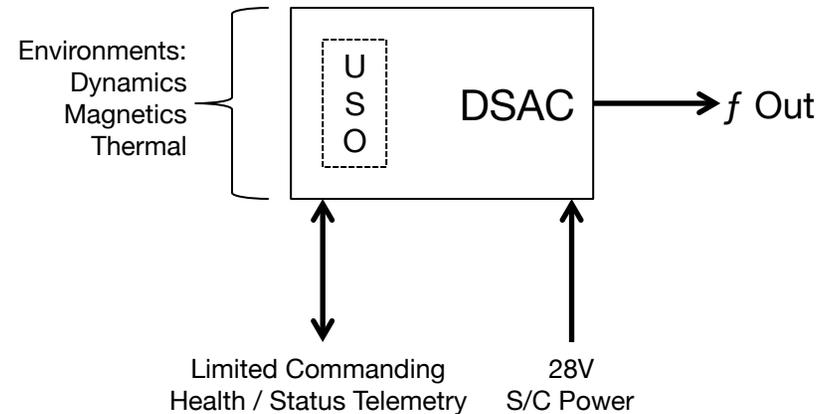
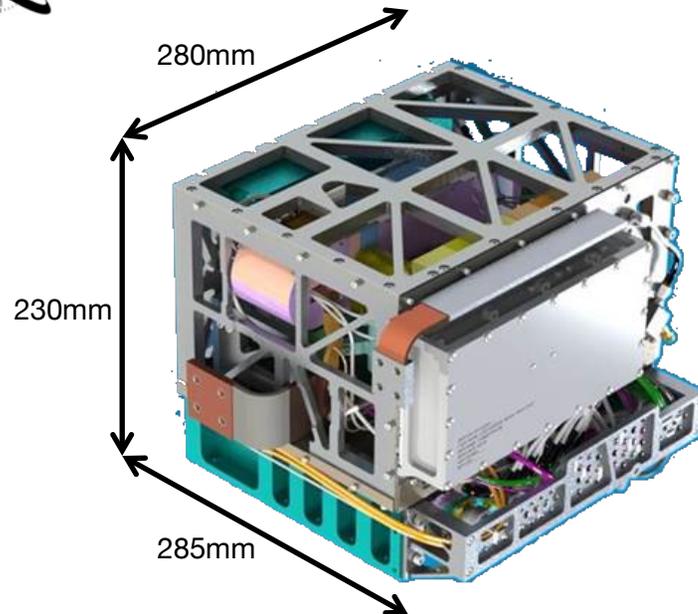


## Implementation Uses

| Deep Space Navigation  | Science   | Deep Space Timing  | Autonomy   | Near Space Navigation/Timing  |
|--|---|--|--|---|
| NASA IND/DSN<br>NASA SMD/PSD   | NASA SMD/PSD  | NASA IND/DSN<br>NASA SMD/PSD   | NASA SMD/PSD   | USAF – SMC/GPS<br>USAF – MILSATCOM<br>NRO   |
| <ul style="list-style-type: none"> <li>Multiple Spacecraft Per Aperture at Mars - doubles useful tracking</li> <li>Full use of Ka-band tracking – OD uncertainty at Mars &lt; 1 m (10 x improvement)</li> <li>Outer planets users gain significant tracking efficiency – 15% at Jupiter 25% at Saturn</li> </ul> | <ul style="list-style-type: none"> <li>Enhance gravity science at Mars, GRACE-level determination of long term gravity with one satellite, at Europa, flyby gravity objectives met robustly</li> <li>Enhance planetary occultation science with 10 x better data</li> </ul> | <ul style="list-style-type: none"> <li>Significantly reduce spacecraft timekeeping overhead</li> <li>Improve reliability of critical time-dependent autonomous spacecraft functions</li> <li>Reduce risks to long-term spacecraft hibernation</li> </ul> | <ul style="list-style-type: none"> <li>Enables autonomous radio navigation (robotic and crewed)</li> <li>Enhances EDL and precision landing</li> <li>Key component to autonomous aerobraking</li> <li>Coupled with OpNav, enhances primitive body exploration</li> </ul> | <ul style="list-style-type: none"> <li>Diversifies clock industrial base - enhancing national security</li> <li>Provides needed time accuracy/ stability for next generation secure communications</li> <li>Significant aid to users with compromised GPS visibility – need only 3 in-view to position</li> </ul> |



## Core Technology At Scale – Demo Electronics and Packaging At TRL7



DSAC is a small, low-mass mercury-ion clock suitable for deep space flight with limited interfaces

Depending on intended use, environments, etc...two options for implementation exist:

- Option 1 (DSAC “As-Is”): SWaP (17kg, 56W w/USO), GEVS Dynamics
- Option 2 (DSAC Reduced): Re-package Electronics to reduce SWaP to 10kg, 30W

For both options, analysis of environments and lifetime would be required  
For both options, analysis of short-term (USO-based) stability is required

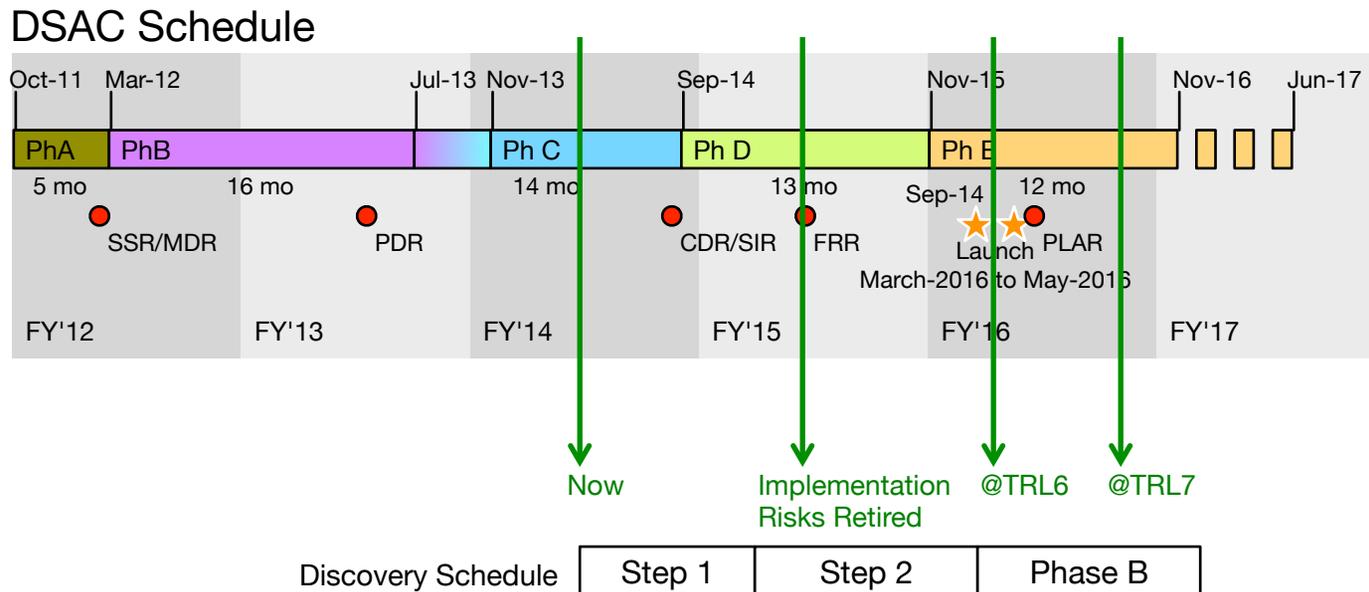


## TRL 7 Definition for DSAC

| TRL 7 defined for the Deep Space Atomic Clock TDM                 |  |
|---|--|
| <b>Capability</b><br><i>performance against need</i>              | Demonstrated capabilities against formal requirements.<br>Assessment of demonstrated capabilities against future mission needs.                            |
| <b>SWaP</b><br><i>Volume, Mass, Power</i>                         | Physics package at SWaP on track for future mission needs.<br>Support Electronics and packaging to meet demonstration flight needs only.                   |
| <b>Model Fidelity</b><br><i>What is flown?</i>                    | Advanced Prototype. Physics Package in the near-final configuration.<br>Support Electronics suitable for demonstration mission only.                       |
| <b>Critical Engineering</b><br><i>How much NRE Remains?</i>       | Designed with flight rules and guidelines. Materials and Process controls.<br>Mostly flight parts, some with flight potential (fly lower screening grade). |
| <b>Environments</b>   | Testing to demonstration mission requirements only.<br>Assessment against future mission needs as can be defined.  |
| <b>Reliability</b><br><i>Lifetime, Parts, Analyses, QA</i>        | Focus on design margins verification and lifetime.<br>Lifetime assessment against 15 year goal.  |
| <b>Validation</b><br><i>How are the LI Requirements Verified?</i> | Verified against formal requirements for the demonstration mission only.<br>Formal assessment against future mission needs that can be identified.         |



## Schedule Alignment with Discovery



*DSAC is phased in advance of Discovery Needs Offering low implementation risk.*



## AO Information Package Contents: Identical to Electra Contents

| DSAC Content                                | Detail                                  |
|---|---|
| <b>General Information</b>                  |   |
| Description of arrangement with NASA        | Agreement Letter                        |
| Points of Contact                           | Project & Technical                     |
| <b>Payload Description</b>                  |   |
| Description of technology development       | Based on DSAC Infusion report           |
| List of advantages offered                  | Based on DSAC Infusion Status Briefings |
| High-level description of components        | No ITAR-restricted Data                 |
| General explanation of operating principles | Yes                                     |
| Image of payload                            | Current DSAC Configuration              |
| Simple block diagram of payload             | Yes, Including Interfaces               |
| Table of key specifications                 | Yes (CBE)                               |
| Table of payload-spacecraft bus interfaces  | Yes (ICD)                               |
| Description of sensitivities/considerations | Yes                                     |



## *ROM Accommodation Cost*

- To a S/C, DSAC is a simple, Power-In, Reference Frequency Out Device
  - Limited Commanding or Health and Status Telemetry Monitoring
- Based on Similar Devices (USO's)
  - The Accommodation Cost is would be based on these considerations:
    - Systems Engineering
      - Selection of USO Characteristics – Base Frequency & Short-term Stability
      - Integration with Telecom System
      - Fault Protection Considerations (if tied to Telecom System)
    - Thermal / Magnetic Environments for optimum performance (if necessary)
      - Only required for the highest performance (E-15 level AD) cases
    - System-level Integration & Test



## *Points of Contact*

- For Program/Project and Implementation Items
  - Allen H. Farrington – DSAC Project Manager, [Allen.H.Farrington@jpl.nasa.gov](mailto:Allen.H.Farrington@jpl.nasa.gov), 818-653-2284
- For DSAC Capabilities, Navigation and Scientific Uses
  - Jill Seubert, [Jill.Tombasco@jpl.nasa.gov](mailto:Jill.Tombasco@jpl.nasa.gov), 818-354-4076
- Jill and Allen are the “entry point” into working more closely with the DSAC/JPL team.
  - Jill and Allen work non-exclusively with all teams while protecting their information from being passed to any other proposal team.
  - The DSAC team and associated SME’s are a small group
    - The SME’s will work for all proposals – briefed on how to prevent COI and how to compartmentalize information about one proposal versus another
    - DSAC Project-Generated Technology and Advancements available to all proposals
      - DSAC is continuing to make progress and advancements and will feed that to all proposers working with us during Steps 1 and 2